### www.msubbu.inEx-1Shell and Tube Heat Exchanger

A shell and tube heat exchanger is to be designed to sub-cool condensate from a methanol condenser from 95°C to 40°C. Flow-rate of methanol is 100,000 kg/h. Brackish water will be used as the coolant, with a temperature rise from 25° to 40°C. A pressure drop of 0.8 bar is permissible on both streams. Allowance should be made for fouling by including a fouling heat transfer coefficient for methanol as 5000  $(W/m^2.°C)$  and 3000  $(W/m^2.°C)$  on the brackish water stream. Thermal conductivity of cupro-nickel alloys is 50 W/m.°C.

The fluid properties at their mean temperatures are given as:

Property	Water (at 32.5°C)	Methanol (at 67.5°C)
Specific heat (kJ/kg. <sup>o</sup> C)	4.2	2.84
Thermal conductivity (W/m. <sup>o</sup> C)	0.59	0.19
Density (kg/m <sup>3</sup> )	995	750
Viscosity (cP)	0.8	0.34

#### The following heat exchanger configuration is available:

- Fixed tube sheet shell and tube exchanger: 1 shell pass and 2 tube passes.
- Brackish water on the tube side and methanol on the shell side.
- 918 number of cupro-nickel tubes of 20 mm OD, 16 mm ID, 4.88 m long each, arranged in triangular pitch, pitch of 1.25 times OD.
- Shell diameter: ID 894 mm, baffle spacing: 356 mm; 25% cut segmental baffles
- Tube side nozzle flanges: 100 NB, 150 psi, Shell side nozzle flanges: 150 NB, 150 psi

(a) Make suitable calculations, to show that the above design is satisfactory. (Start with an approx value for U of  $600 \text{ W/m}^{2.\circ}\text{C}$ ).

(b) Draw to scale the half-sectional elevation and tube-sheet layout of the heat exchanger. Mark the salient parts of the exchanger.

# **Design Calculations:**

Coolant is corrosive, so assign to tube-side.

Heat capacity methanol = 2.84 kJ/kg°C  
Heat load = 
$$\frac{100,000}{3600}$$
 × 2.84(95 – 40) = 4340 kW  
Heat capacity water = 4.2 kJ/kg°C  
Cooling water flow =  $\frac{4340}{4.2(40 - 25)}$  = 68.9 kg/s  
 $\Delta T_{\rm lm} = \frac{(95 - 40) - (40 - 25)}{\ln \frac{(95 - 40)}{(40 - 25)}} = 31°C$ 

Use one shell pass and two tube passes

$$R = \frac{95 - 40}{40 - 25} = 3.67$$
$$S = \frac{40 - 25}{95 - 25} = 0.21$$

From Figure 12.19

$$F_r = 0.85$$
$$\Delta T_m = 0.35 \times 31 = 26^{\circ} \text{C}$$

From Figure 12.1

$$U = 600 \text{ W/m}^{2} \text{°C}$$

Provisional area

$$A = \frac{4340 \times 10^3}{26 \times 600} = 278 \text{ m}^2$$

Choose 20 mm o.d., 16 mm i.d., 4.88-m-long tubes  $(\frac{3}{4}$  in. × 16 ft), cupro-nickel.

Allowing for tube-sheet thickness, take

$$L = 4.83 \text{ m}$$

$$L = 4.60 \text{ m}$$

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$$\frac{1}{2}$$
 or tubes  $-\frac{1}{0.303} - \frac{210}{1}$   
relatively clean use 1.25 triangula

As the ar pitch.

Bundle diameter  $D_b = 20$ 

 $\left(\frac{918}{0.249}\right)$ 

= 826 mm

1/2.207

Use a split-ring floating head type.

From Figure 12.10, bundle diametrical clearance = 68 mm,

shell diameter,  $D_s = 826 + 68 = 894$  mm.

$$\frac{1}{10000} = \frac{1}{0.303} = \frac{1}{1000}$$

$$0.303 - 100$$

Number of tubes = 
$$\frac{0.303}{0.303} = \frac{918}{1000}$$

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Number of tubes = 
$$\frac{0.303}{0.303} = \frac{918}{1000}$$

per of tubes 
$$= \frac{0.303}{0.303} = \frac{918}{1000}$$

$$U = 600 \text{ W/m}^2$$

#### **Tube Side Heat Transfer Coefficient**

Mean water temperature = 
$$\frac{40 + 25}{2} = 33^{\circ}$$
C  
Tube cross-sectional area =  $\frac{\pi}{4} \times 16^2 = 201 \text{ mm}^2$   
Tubes per pass =  $\frac{918}{2} = 459$   
Total flow area =  $459 \times 201 \times 10^{-6} = 0.092 \text{ m}^2$   
Water mass velocity =  $\frac{68.9}{0.092} = 749 \text{ kg/s m}^2$   
Density water =  $995 \text{ kg/m}^3$   
Water linear velocity =  $\frac{749}{995} = 0.75 \text{ m/s}$   
 $\frac{h_i d_i}{k_f} = j_k Re P r^{0.33} \left(\frac{\mu}{\mu_w}\right)^{0.14}$   
Viscosity of water =  $0.8 \text{ mNs/m}^2$   
Thermal conductivity =  $0.59 \text{ W/m}^\circ$ C  
 $Re = \frac{\rho u d_i}{\mu} = \frac{995 \times 0.75 \times 16 \times 10^{-3}}{0.8 \times 10^{-3}} = 14,925$   
 $Pr = \frac{C_p \mu}{k_f} = \frac{4.2 \times 10^3 \times 0.8 \times 10^{-3}}{0.59} = 5.7$   
Neglect  $\left(\frac{\mu}{\mu_w}\right)$   
 $\frac{L}{d_i} = \frac{4.83 \times 10^3}{16} = 302$   
From Figure 12.23,  $j_h = 3.9 \times 10^{-3}$   
 $h_i = \frac{0.59}{16 \times 10^{-3}} \times 3.9 \times 10^{-3} \times 14,925 \times 5.7^{0.33} = 3812 \text{ W/m}^2 \,^\circ \text{C}$ 

Shell Side Heat Transfer Coefficient:

$$A_s = \frac{(p_t - d_o)D_s l_B}{p_t}$$

$$G_s = rac{W_s}{A_s}$$
 where  $W_s$  = fluid flow-rate on the shell-side, kg/s,  
 $u_s = rac{G_s}{
ho}$   $ho$  = shell-side fluid density, kg/m<sup>3</sup>.

For an equilateral triangular pitch arrangement:

$$d_e = \frac{4\left(\frac{p_t}{2} \times 0.87p_t - \frac{1}{2}\pi \frac{d_o^2}{4}\right)}{\frac{\pi d_o}{2}} = \frac{1.10}{d_o}(p_t^2 - 0.917d_o^2)$$

where  $d_e =$  equivalent diameter, m.

$$Re = \frac{G_s d_e}{\mu} = \frac{u_s d_e \rho}{\mu}$$
  
Choose baffle spacing  $= \frac{D_s}{5} = \frac{894}{5} = 178$  mic.  
Tube pitch  $= 1.25 \times 20 = 25$  mm

The spacing 
$$=\frac{1}{5} = \frac{1}{5} = 1.76$$
 mgc.  
Tube pitch  $= 1.25 \times 20 = 25$  mm  
Cross-flow area  $A_s = \frac{(2.5 - 20)}{25} 894 \times 178 \times 10^{-6} = 0.032$  m<sup>2</sup>  
Mass velocity,  $G_S = \frac{100,000}{3600} \times \frac{1}{0.032} = 868$  kg/s m<sup>2</sup>  
Equivalent diameter  $d_e = \frac{1.1}{20} (25^2 - 0.917 \times 20^2) = 14.4$  mm

Mean shell side temperature =  $\frac{95 + 40}{2} = 68^{\circ}C$ 

Methanol density =  $750 \text{ kg/m}^3$ 

Viscosity =  $0.34 \text{ mNs/m}^2$ 

Heat capacity = 2.84 kJ/kg°C

Thermal conductivity = 0.19 W/m°C

$$Re = \frac{G_s d_e}{\mu} = \frac{868 \times 14.4 \times 10^{-3}}{0.34 \times 10^{-3}} = 36,762$$

$$Re = \frac{C_{pre}}{\mu} = \frac{0.34 \times 10^{-3}}{0.34 \times 10^{-3}} = 36,762$$
$$Pr = \frac{C_{p}\mu}{k_{f}} = \frac{2.84 \times 10^{3} \times 0.34 \times 10^{-3}}{0.19} = 5.1$$

Choose 25 per cent baffle cut, from Figure 12.29

$$i_h = 3.3 \times 10^{-3}$$

Without the viscosity correction term

$$h_s = \frac{0.19}{14.4 \times 10^{-3}} \times 3.3 \times 10^{-3} \times 36,762 \times 5.1^{1/3} = 2740 \text{ W/m}^{2} \text{°C}$$

#### **Overall Heat Transfer Coefficient:**

$$\frac{1}{U_o} = \frac{1}{h_o} + \frac{1}{h_{od}} + \frac{d_o \ln\left(\frac{d_o}{d_i}\right)}{2k_w} + \frac{d_o}{d_i} \times \frac{1}{h_{id}} + \frac{d_o}{d_i} \times \frac{1}{h_i}$$

where  $U_o$  = the overall coefficient based on the outside area of the tube, W/m<sup>2</sup>°C,

 $h_o$  = outside fluid film coefficient, W/m<sup>2</sup>°C,

 $h_i$  = inside fluid film coefficient, W/m<sup>2</sup>°C,

 $h_{od}$  = outside dirt coefficient (fouling factor), W/m<sup>2</sup>°C,

 $h_{id}$  = inside dirt coefficient, W/m<sup>2</sup>°C,

 $k_w$  = thermal conductivity of the tube wall material, W/m°C,

 $d_i$  = tube inside diameter, m,

 $d_o$  = tube outside diameter, m.

$$\frac{1}{U_o} = \frac{1}{2740} + \frac{1}{5000} + \frac{20 \times 10^{-3} \ln\left(\frac{20}{16}\right)}{2 \times 50} + \frac{20}{16} \times \frac{1}{3000} + \frac{20}{16} \times \frac{1}{3812}$$
$$U_o = \underline{738 \text{ W/m}^2 \circ \text{C}}$$

which is well above the assumed value of 600  $W/m^2$ . C. Hence the given configuration is satisfactory.

#### **Pressure Drop Calculations:**

#### Tube-side

From Figure 12.24, for Re = 14,925

$$j_f = 4.3 \times 10^{-3}$$

Neglecting the viscosity correction term

$$\Delta P_t = 2\left(8 \times 4.3 \times 10^{-3} \left(\frac{4.83 \times 10^3}{16}\right) + 2.5\right) \frac{995 \times 0.75^2}{2}$$
  
= 7211 N/m<sup>2</sup> = 7.2 kPa (1.1 psi)

low, could consider increasing the number of tube passes.

## Shell side

Linear velocity = 
$$\frac{G_s}{\rho} = \frac{868}{750} = 1.16$$
 m/s

From Figure 12.30, at Re = 36,762

$$j_f = 4 \times 10^{-2}$$

Neglect viscosity correction

$$\begin{split} \Delta P_s &= 8 \times 4 \times 10^{-2} \left(\frac{894}{14.4}\right) \left(\frac{4.83 \times 10^3}{178}\right) \frac{750 \times 1.16^2}{2} \\ &= 272,019 \text{ N/m}^2 \\ &= 272 \text{ kPa (39 psi) too high,} \end{split}$$

could be reduced by increasing the baffle pitch. Doubling the pitch halves the shell-side velocity, which reduces the pressure drop by a factor of approximately  $(1/2)^2$ 

$$\Delta P_s = \frac{272}{4} = 68 \text{ kPa} (10 \text{ psi}), \text{ acceptable}$$

This will reduce the shell-side heat-transfer coefficient by a factor of  $(1/2)^{0.8}(h_o \propto Re^{0.8} \propto u_s^{0.8})$ 

$$h_0 = 2740 \times (\frac{1}{2})^{0.8} = 1573 \text{ W/m}^{2} \circ \text{C}$$

This gives an overall coefficient of 615 W/m<sup>2</sup> °C – still above of 600 W/m<sup>2</sup> °C. assumed value