Heat Transfer Convection - Correlations for Heat Transfer Coefficient

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Objectives

• To give an overview about the variables of importance in convective heat transfer



- To understand the significance of various dimensionless numbers in convection.
- To understand the various empirical correlations used and their specific conditions of usage in convective heat transfer.



Hydrodynamic Boundary Layer



Figure : Flow development in pipe

Entry Length:

Laminar flow:

Turbulent flow:

$$\left(\frac{L}{D}\right)_{e,h} \approx 0.05 \text{ Re}$$

 $\left(\frac{L}{D}\right)_{e,h} > 10$

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Critical Reynolds Number

Critical Reynolds number approximates the location where the flow transitions from laminar to turbulent flow.

 ${
m Re}_{x,c}pprox 10^5$ external (flat plate) flow ${
m Re}_{D,c}pprox 10^3$ internal (duct) flow



Thermal Boundary Layer



Figure : Thermally developing flow

Thermal Entry Length: for laminar flow,

$$\left(\frac{L}{D}\right)_{e,t} \approx 0.05 \text{ Re Pr}$$









Comparing thermal boundary layer with hydrodynamic boundary layer, it can be said that if Pr > 1, hydrodynamic boundary layer grows more rapidly.

 $\Pr > 1, \quad L_{e,h} < L_{e,t}$ $\implies \delta > \delta_T$ at any section

Inverse is true for Pr < 1.

It is significant that the Prandtl number for most gases are sufficiently close to unity, and the hydrodynamic and thermal boundaries are of similar extent.

When $\Pr > 1$, $\delta > \delta_t$, and when $\Pr < 1$, $\delta < \delta_t$. High viscosity leads to a thick momentum boundary layer, and high thermal diffusivity leads to a thick thermal boundary layer.



Thermally Fully Developed Conditions

Since the existence of convective heat transfer between the surface and the fluid dictates that the fluid temperature must continue to change with x, one might question whether fully developed conditions ever can be reached. The situation is certainly different from the hydrodynamic case, for which $(\partial v / \partial x) = 0$ in the fully developed region. In contrast, if there is heat transfer, $(\partial T/\partial x)$ at any radius is not zero. Introducing a dimensionless temperature $(T_w - T)/(T_w - T_m)$, condition for which this ratio becomes independent of x are known to exist. Although the temperature profile T(r) continues to change with x, the relative shape of the profile does not change and the flow is said to be fully developed. Instead of $(\partial T / \partial x) = 0$, the condition is

$$\frac{d}{dx}\left[\frac{T_w(x) - T(r, x)}{T_w(x) - T_m(x)}\right] = 0$$

i.e., $(T_w - T)$ changes in the same way as $(T_w - T_m)$.

Tube surface condition: either uniform wall temperature condition (UWT) condition, or a uniform wall heat flux condition (UWH). For $T_w =$ constant, i.e., $(dT_w/dx) = 0$: If a phase change were occurring at outer surface.

For $q_w = h(T_w - T_m) =$ constant: If the tube wall were heated electrically or, if the outer surface were uniformly irradiated.



Constant Wall Temperature



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Constant Wall Heat Flux



By assumption, $q_w = h(T_w - T_m) = \text{ constant.}$ For thermally developed flow, h = constant. Then,

$$\frac{d}{dx}(q) = \frac{d}{dx}[h(T_w - T_m)] = 0$$

i.e.,

$$0 = h\left(\frac{dT_w}{dx} - \frac{dT_m}{dx}\right)$$

Convection

 $\frac{dT_w}{dx} = \frac{dT_m}{dx}$

Variation of Temperature with Distance Constant Wall Heat Flux



 $\frac{dT_m}{dx} = \frac{q_w P}{\dot{m}C_P} = \text{ constant}$

where P is the perimeter of wall. Integrating from x = 0, we get

$$T_m(x) = T_{m,i} + \frac{q_w P}{\dot{m}C_P} x \implies$$
 Linear variation

Variation of Temperature with Distance Constant Wall Temperature

 T_w is constant.

$$\frac{T_w - T_m(x)}{T_w - T_{m,i}} = \exp\left[-\left(\frac{Ph}{\dot{m}C_P}\right)x\right]$$

$$q = hA\Delta T_{\rm Im}$$

where

 $\Delta T_{\rm Im} = \text{Log Mean Temperature Difference (LMTD)} = \frac{\Delta T_o - \Delta T_i}{\ln(\Delta T_o / \Delta T_i)}$



Note: in all equations evaluate fluid properties at the film temperature (T_f) defined as the arithmetic mean of the surface and free-stream temperatures unless otherwise stated.

$$T_f = \frac{T_S + T_\infty}{2} = \frac{T_w + T_m}{2}$$



Local and Average Heat Transfer Coefficients







Local and Average Heat Transfer Coefficients (contd..)



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- Friction factor and the heat transfer coefficient remain constant in fully developed laminar or turbulent flow, since the velocity and normalized temperature profiles do not vary in the flow direction.
- Laminar and turbulent boundary layers have different heat transfer characteristics turbulent mixing typically increases heat transfer.



When a surface is subjected to external flow, the problem involves both natural and forced convection. The relative importance of each mode of heat transfer is determined by the value of the coefficient Gr/Re^2 :

- Natural convection effects are negligible if $Gr/Re^2 \ll 1$.
- Free convection dominates and the forced convection effects are negligible if ${\rm Gr}/{\rm Re}^2 \gg 1$.
- Both effects are significant and must be considered if $Gr/Re^2 = 1$ (mixed convection).





(a) Natural convection $(Gr_L / Re_L^2 >> 1)$



(b) Forced convection $(Gr_L / Re_L^2 << 1)$



(c) Mixed convection $(Gr_L / Re_L^2 \approx 1)$

FIGURE

The relative importance of convection heat transfer regimes for flow near a hot sphere.

Dimensionless Numbers

Dimensionless number	Formula	Importance
Reynolds number (Re)	$\frac{Dv\rho}{\mu}$	$Re = \frac{inertial force}{viscous force}$ Re < 2000, flow is laminar.
Prandtl number (Pr)	$\frac{C_{p\mu}}{k}$	$\Pr = \frac{(\mu/\rho)}{(k/\rho C_p)} = \frac{\text{molecular diffusivity of momentum}}{\text{molecular diffusivity of heat}}$
Nusselt number (Nu)	$\frac{hL}{k}$	$\begin{split} \mathrm{Nu} &= \frac{h\Delta T}{k\Delta T/L} = \frac{\mathrm{heat\ transfer\ by\ convection}}{\mathrm{heat\ transfer\ by\ conduction}} \mathrm{where} \\ \Delta T \ \mathrm{is\ temperature\ difference\ between\ the\ wall\ surface} \\ \mathrm{and\ fluid\ temperatures.\ Nusselt\ number\ equal\ to\ unity} \\ \mathrm{implies\ that\ there\ is\ no\ convection\ -\ the\ heat\ transfer} \\ \mathrm{is\ purely\ by\ conduction\ across\ the\ layer\ of\ thickness\ L.} \end{split}$
Grashof number (Gr)	$\frac{g\beta L^3(T_w - T_\infty)}{(\mu/\rho)^2}$	$ \begin{array}{l} {\rm Gr} = \frac{({\rm buoyancy\ force})^2}{({\rm viscous\ force})^2} \beta = {\rm volumetric\ coefficient} \\ {\rm of\ thermal\ expansion\ of\ the\ fluid\ (unit:\ 1/^{\circ}{\rm C})} \end{array} $
Parameter Gr/Re ²	$\frac{\mathrm{Gr}}{\mathrm{Re}^2}$	$\label{eq:rescaled} \begin{array}{l} \displaystyle \frac{Gr}{Re^2} = \frac{(buoyancy\ force)^2}{(inertial\ force)^2} & \mbox{It is a measure of the relative importance of the free convection in relation to forced convection. If (Gr/Re^2) \ll 1,\ flow\ is\ primarily by\ forced\ convection. \end{array}$

Dimensionless Numbers (contd..)

Dimensionless number	Formula	Importance
Rayleigh number (Ra)	Gr.Pr	Used in place of Grashof number to correlate heat transfer in free convection.
Peclet number (Pe)	Re.Pr	Used in place of Reynolds number to correlate heat transfer in forced convection.
Stanton number (St)	$\frac{\mathrm{Nu}}{\mathrm{Re}.\mathrm{Pr}}$	$\begin{split} \mathrm{St} &= \frac{h\Delta T}{\rho C_p v \Delta T} \text{The numerator represents the heat} \\ \mathrm{flux \ to \ the \ fluid, \ and \ denominator \ represents \ the \ heat} \\ \mathrm{transfer \ capacity \ of \ the \ fluid \ flow.} \end{split}$
Graetz number (Gz)	$\frac{\text{Re} \cdot \text{Pr}}{L/D}$	Graetz number is a correlating parameter in thermally developing flow.



Laminar flow:

$$Nu_x = \frac{h_x x}{k} = 0.332 \text{ Re}_x^{0.5} \text{ Pr}^{1/3}$$
 Pr > 0.6

Turbulent flow:

$$Nu_{\rm x} = \frac{h_{\rm x} x}{k} = 0.03 \; {\rm Re}_{\rm x}^{0.8} \; {\rm Pr}^{1/3} \qquad \begin{array}{c} 0.6 \le {\rm Pr} \le 60 \\ 5 \times 10^5 \le {\rm Re}_{\rm x} \le 10^7 \end{array}$$

Note:

For laminar flow, h_x is proportional to $\operatorname{Re}_x^{0.5}$ and thus to $x^{-0.5}$. For turbulent flow, h_x is proportional to $x^{-0.2}$.



Flow over Flat Plate Average Heat Transfer Coefficient

Laminar flow:

$$Nu_L = \frac{hL}{k} = 0.664 \text{ Re}_L^{0.5} \text{ Pr}^{1/3}$$
 Pr > 0.6



Forced Convection - Laminar Flow through Circular Pipe

Convection heat transfer associated with laminar flow in a circular pipe (Re < 2100) is less common than with turbulent flow. The correlations are simple, with Nusselt number being constant for fully developed flow (both hydrodynamically and thermally).

For	uniform	wall	heat flux:	Nu=4.36
For	uniform	wall	temperature:	Nu = 3.66

The L/D ratio for the entrance length required to reach fully developed flow is larger for laminar flow than for turbulent flow.



Forced Convection - for Turbulent Flow inside Circular Pipe

Dittus-Boelter equation (1930)

$$\mathsf{Nu} = 0.023 \; \mathsf{Re}^{0.8} \; \mathsf{Pr}^n$$

where

$$n = \begin{cases} 0.3 \text{ for cooling of the fluid } (T_s < T_{\infty}) \\ 0.4 \text{ for heating of the fluid } (T_s > T_{\infty}) \end{cases}$$

The Dittus-Boelter equation is valid for smooth pipes and for

$$0.6 \le \Pr \le 160$$
 Re $\ge 10,000$ $\frac{L}{D} \ge 10$



Forced Convection - for Turbulent Flow inside Circular Tube (contd..)

The Dittus-Boelter equation is recommended only for rather small temperature differences between the bulk fluid and the pipe wall. A few years later, in 1936, Sieder and Tate proposed the following equation to accommodate larger temperature differences:

Nu = 0.023 Re^{0.8} Pr^{1/3}
$$(\mu/\mu_w)^{0.14}$$

where

 μ = viscosity of the fluid at the fluid bulk temperature μ_w = viscosity of the fluid at the pipe wall temperature Valid for:

$$0.7 \le \Pr \le 16,700$$
 Re $\ge 10,000$

Both the Dittus-Boelter and Seider-Tate equations are still in widespread use.

 $\frac{L}{D} \ge 10$

Free Convection



Figure 9: Flow Patterns for Various Conditions in Free Convection (T_s = temperature of plate surface; T_{∞} = temperature of bulk fluid)



Free Convection (contd..)



Figure 10: Fluid Contained between Two Horizontal Plates



Free Convection - Correlations

• Isothermal Vertical Plate

$$Nu = \begin{cases} 0.59 \text{ Ra}^{1/4} & 10^4 < \text{Ra} < 10^9 & (\text{laminar flow}) \\ 0.1 \text{ Ra}^{1/3} & 10^9 < \text{Ra} < 10^{13} & (\text{turbulent flow}) \end{cases}$$

• Isothermal Horizontal Plate

- Hot plate facing upwards:

$$Nu = \begin{cases} 0.54 \text{ Ra}^{1/4} & 10^4 < \text{Ra} < 10^7 \\ 0.15 \text{ Ra}^{1/3} & 10^7 < \text{Ra} < 10^{11} \end{cases}$$

- Hot plate facing downwards:

$$Nu = 0.27 \text{ Ra}^{1/4}$$
 $10^7 < \text{Ra} < 10^{11}$